AVIONICS SYSTEM

CLEAR INFORMATION SIMPLER DECISION





ASA

funke.

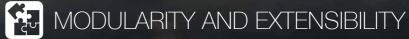
INTEGRATED AVIONICS SYSTEM





© CONDITION-BASED MAINTENANCE





Instant Insight



Intuitive colour codes indicate the status of the aircraft at a glance. Simple graphics immediately convey key information. With ergonomically designed knobs and excellent workmanship, our instruments are to make you enjoy the flight.



















Some parameters are outside acceptable limits.

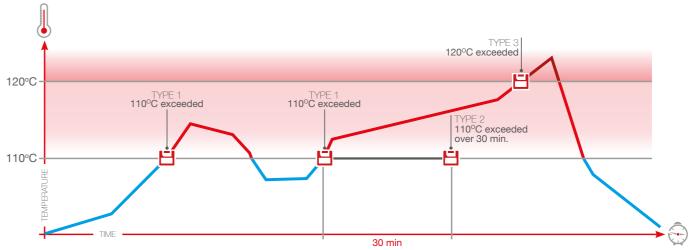






The system continuously analyses flight and engine performance parameters to detect instances of exceeding the normal operating range. All deviations are automatically recorded together with flight time and HOOBS without engaging the pilot's attention. This way, the system becomes the pilot's personal assistant and facilitates log book keeping.



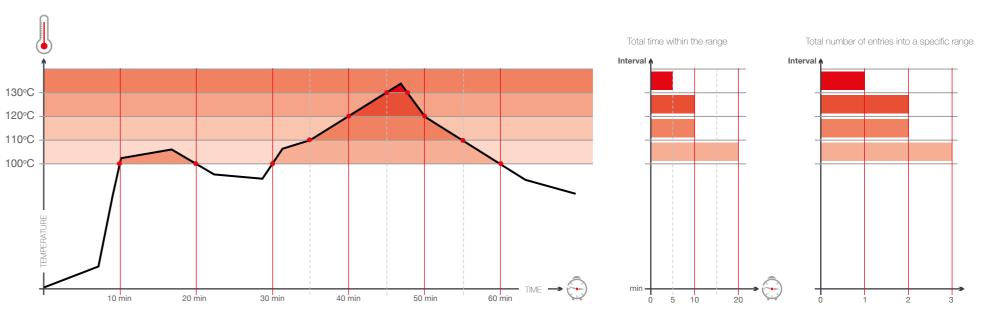


The system distinguishes between various types of exceedance events. The events may be just instances of exceeding a threshold value detected while monitoring a particular parameter. However, they may be a result of complex conditions where not only the fact of exceedance, but also its duration are of interest. Therefore, the system remembers four last events of each type, providing necessary data.



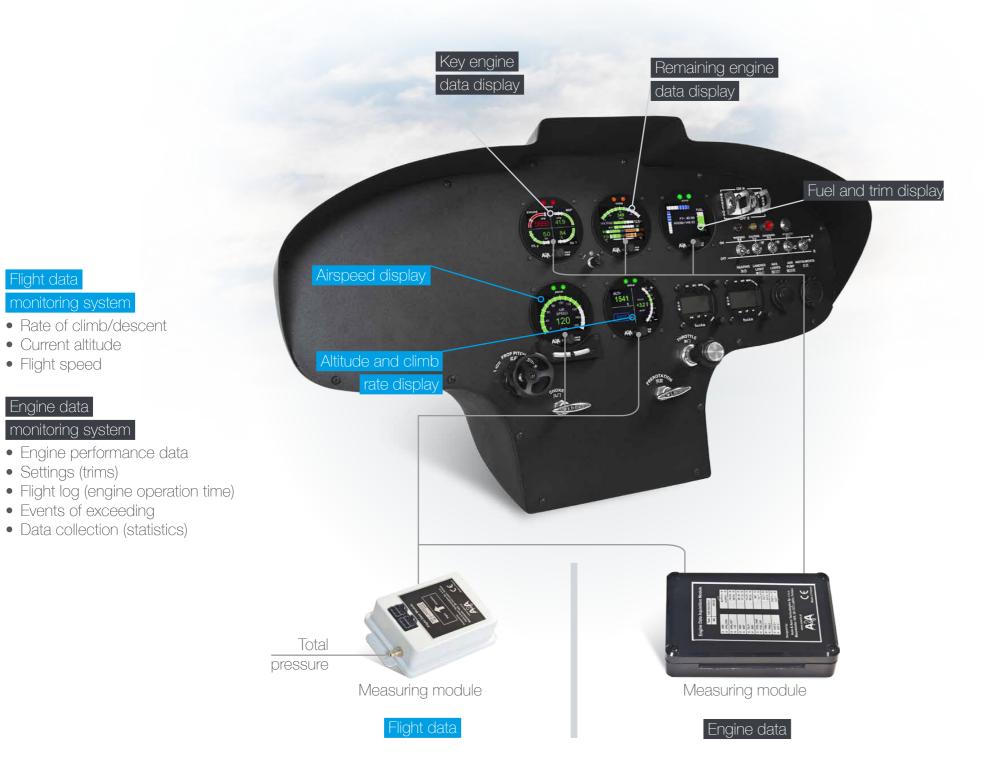


It is better to prevent than to cure. The system monitors key operating parameters of both the engine and the aircraft. It records how often, to what extent and for how long parameter values were outside the desired range. These records are an invaluable source of operation history data that help detect unwelcome trends before they develop into failure.



The idea of monitoring consists in splitting the range between the desired level and exceedance level of a parameter into sub-ranges. Then the system counts the number of instances of the parameter's value entering each sub-range and measures total time of the parameter's value staying within a specific sub-range.





Rate of climb/descent

 Current altitude Flight speed

monitoring system

Settings (trims)

Events of exceeding

• Engine performance data

Data collection (statistics)

Engine data



The Integrated Avionics System is designed for ultralights and LSA (Light Sport Aircraft).

It consists of two sub-systems: the flight data monitoring and the engine performance data monitoring. The sub-systems are possible to be used separately, but combining them provides synergetic effect. The modular design allows the user to add other compatible elements in the future.

HIGH QUALITY WORKMANSHIP

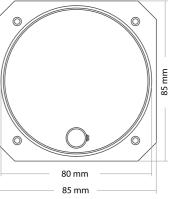
The instruments' anodised aluminium casing ensures high durability and aesthetics. The anti-reflective glass and brightness control makes them convenient to use in any lighting conditions.

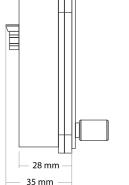
ENERGY EFFICIENCY

The key design objective was to minimise the system's energy demand. The total consumption of electrical energy does not exceed **6,5 W.**

SMALL SIZE

The instruments' size was optimized to provide a comfortable display in a standard diameter (80mm) casing that is only **28 mm** deep.











The system is designed to measure altitude, vertical speed and airspeed.

It consists of three components:

- AUX AVS 02 Measurement module
- EEM MOT 01 F07.01 Display
- EEM MOT 01 F06.01 Display

The AUX AVS 2 measurement module is equipped with two pressure gauges to detect static pressure and total pressure. The static pressure is measured inside the module's casing (the module ambient pressure), whereas the total pressure requires connection to the Pitot tube. The module contains an integrated system for measuring the magnetic direction, roll, pitch, and yaw, and axial acceleration. This enables the pilot to determine flight direction and to analyse exceeding events under the aircraft operating conditions. The reference pressure used to determine the flight altitude is set up by means of the knob.

Item	PARAMETER	MEASURING RANGE	ACCURACY
1	Altitude – determined based on the measured static pressure and the reference pressure set up by the pilot	0 – 3000 metres	±8m
2	Vertical speed – determined on the basis of altitude changes	±12 m/s	±0,2 m/s
3	Airspeed – determined based on the dynamic pressure (measured differentially between the total pressure in the Pitot tube and the static pressure in the pilot's cabin)	25 – 250 km/h	±5 km/h

¹ in development

MEASURING MODULE	AUX AVS 2
Dimensions	74 x 120 x 32 mm
Weight	85 g
Casing material	ABS
Voltage	12 V
Current	<12mA
Input	Pitot tube (total pressure)





Avionics Instruments



MONITOR	EEM MOT 01
Dimensions	85 x 85 x 67 mm
Installation hole	Ø 80 mm
Weight	175 g
Casing material	Anodised aluminium, acrylic glass with anti-reflective coating
Voltage	12 V
Current	<85mA



SOFTWARE VERSION	F06.01
Displayed parameters	1) Altitude above sea level (m, ft) 2) Vertical speed (m/s, ft/min) 3) Reference pressure (hPa)
Parameters set up by means of the handwheel	Reference pressure
Alarms and warnings	



SOFTWARE VERSION	F07.01
Displayed parameters	Airspeed (km/h, knt)
Alarms and warnings	





The system is compatible with the following engine types: Rotax TM 912UL, 912ULS, 914 and their modifications. It allows the user to measure, display and analyse the basic engine performance parameters.

The system consists of four components:

- EMS DAQ R2 Measuring module
- EEM MOT 01 F02.01 Display
- EEM MOT 01 F08.01 Display
- EEM MOT 01 F09.01 Display

LP.	PARAMETER	MEASURING RANGE	ACCURACY
1	Engine speed	0 ÷ 6500 obr/min	± 10 rpm
2	Rotor speed	0÷500 obr/min	± 1 rpm
3	Inlet air pressure	0÷4 bar	± 0.1 bar
4	Oil pressure	0÷10 bar	± 2°C
5	Oil temperature	-20÷200°C	± 2°C
6	Cylinder head temperature (two cylinders)	-20÷200°C	± 2°C
7	Exhaust gas temperature (two cylinders)	-50÷950°C	±5°C
8	Inlet air temperature	-20÷200°C	± 2°C
9	Fuel level	8÷120 l	± 1
10	Trim (pitch and roll signal)	0÷10 V	± 0.1V
11	Engine operation time		±18
12	Total engine operation time (HOOBS)		± 1 min

MEASURING MODULE	EMS DAQ R02.01
Dimensions	80 x 120 x 28 mm
Weight	160 g
Casing material	ABS
Voltage	12 V
Current	<80mA1
Location	Engine compartment
Compatibility with the engines	ROTAX 912 ULS, ROTAX 912 UL, ROTAX 914

¹ including the sensors' power consumption



Measuring module EMS DAQ R2



Engine Instruments



SOFTWARE VERSION	F02.01
Displayed parameters	1) Engine speed (rpm) 2) Inlet air pressure (bar, inHg) 3) Oil pressure (bar) 4) Oil temperature (°C)
Alarms and warnings	For each parameter



SOFTWARE VERSION	F08.01
Displayed parameters	1) Rotor speed (rpm) 2) Voltage (V) 3) Engine temperature CHT x2 (°C) 4) Exhaust gas temperature EGT x2 (°C)
Alarms and warnings	For rotor speed and voltage
	For temperatures



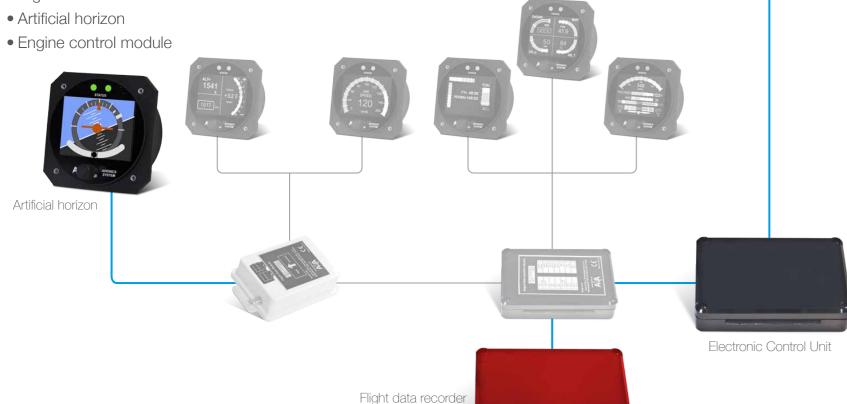
STATUS	SOFTWARE VERSION	F09.01
FT= 48:00 HOO8S=148:03	Displayed parameters	1) Fuel level (litre) 2) Trim: pitch, roll 3) Engine operation time (HH:MM) 4) Total engine operation time (HH:MM)
A AVONCS	Alarms and warnings	



The modular design of the system allows the user to add new compatible elements in the future.

Upcoming modules:

Flight data recorder



Engine Control

Unit Display

About the Company



For over 15 years we have combined knowledge, passion and responsibility to create reliable solutions for the aerospace and automotive industries. Experience and knowledge from many disciplines allow us to understand the nature of a particular project and meet your needs in a responsible manner.

Auto & Aero Technologies is an innovative company creating modern solutions for demanding applications. We specialize in electronics design and software development, mechatronics, hydraulics, pneumatics and mechanical systems design.

Our team designed and made the following aviation solutions:

- Electronic fuel injection system for the ASz-62IR family of aircraft engines
- Electronic ignition system for the ASz-62IR family of aircraft engines
- Electronic control system for the PZL-200 aircraft engine
- Electronic fuel injection system for the iCOMPRESS aircraft engine
- Computer controlled test stand used in research, development, production and certification process of the ASz-62IR aircraft engine family
- Computer controlled test stand used in research, development and production of the PZL-200 aircraft engine

Our partners:

- WSK PZL Kalisz S.A.
- Aviation Artur Trendak
- Institute of Aviation in Warsaw



